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Behaviour of 316L Stainless Steel containing Corrosion Pits under Cyclic Loading

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Abstract:

The environmental performance of 316L grade stainless steel, in the form of tensile specimens containing a single corrosion pit with various geometries, under cyclic loading in aerated chloride solutions is investigated in this work. Results from environmental tests were compared and contrasted with those obtained using Finite Element Analysis (FEA). Fractography of the failed specimens obtained from experiments revealed that fatigue crack initiation took place at the base of the shallow pit. The crack initiation shifted towards the shoulder and the mouth of the pit as the pit depth increased. This process is well predicted by FEA, as the strain contour maps show that strain is the highest around the centric strip of the pit. However, for shallow pits, local strain is uniformly distributed around that strip but begins to concentrate more towards the shoulder and the mouth region as the pit increases in depth.

Keywords: Corrosion pits, FEA, fatigue, pit-crack transition, 316L stainless steel

1. Introduction

Pitting corrosion is a localised damage mechanism that can occur on the surface of many metal alloys. The process can be defined as the accelerated local dissolution of metal, occurring as a result of the breakdown of the otherwise protective passive film present on the metal surface, thus leading to small cavities ('pits'). Pitting is more difficult to detect, predict and design against in comparison to uniform corrosion and as such is considered more dangerous. The morphology of pits may vary considerably, in terms of geometric shape and depth-width (aspect) ratio. Pits propagate with complex directionality, as a function of local stress, microstructure and pit chemistry. Pitting corrosion is particularly problematic for structures operating in marine environments, due to the presence of high concentrations of metal chlorides (NaCl, MgCl₂ and CaCl₂), which are aggressive to steel alloys. Pits can act as local stress raisers in a material and hence can participate in the failure of a material through fatigue and/or stress corrosion cracking (SCC) mechanisms. Consequently, research focussed on defining the range of conditions for which pitting might occur during operational life, allied with determination of the probability of the pit to crack transition, has significant potential to improve material selection at the design stage and structural integrity in general.

Pitting corrosion fatigue life consists of pit initiation, pit growth, crack initiation and crack growth to final failure [1,2]. The available fatigue crack initiation models [2–8] are in fact based on the crack propagation approach, rather than initiation, as they assume a pit to behave as a small crack and use transition criterion to predict the pit-to-crack transition. This criterion uses the stress intensity factor as described by linear elastic fracture mechanics (LEFM), which is valid only when the yield zone ahead of crack is small. Due to the possible large size of the plastic zone around corrosion pits, the possibility of pit local stress to be the driving force for crack initiation, rather than stress intensity factor, is significant. More specifically,

the small-scale yielding and linear elastic assumption may be unrealistic for crack initiation prediction models as applied to pits [9]. In this paper, we assume that a pit can be considered as a notch, i.e. a stress raiser, rather than a crack.

For fatigue life prediction of a material, the pit-to-crack transition stage possesses significant challenges. The transition from a pit-to-crack is not well understood and the early stages of crack development lacks significant insight, leading to uncertainty of both the location of crack nucleation from the pit and the conditions producing crack nucleation. Previous studies [6] have generally assumed that cracks initiate from the base of the corrosion pit, but Turnbull et. al. [10,11] has shown that this is not always the case, as these workers found that the majority of the cracks initiated near the mouth of the corrosion pits.

In the work described herein, we have performed corrosion fatigue tests and supplementary FEA to compare and contrast with the results of tests to better interpret the pit-to-crack process. Single corrosion pits of varying aspect ratios (depth-to-width) were grown on lab specimens made of 316L grade stainless steel (SS 316L) and exposed them to aerated saline solutions under cyclic loads. Surface analysis was carried out on the failed specimens using a Scanning Electron Microscope (SEM) to determine the location of the crack initiation site and crack path for a given geometry of corrosion pit. The local stress field at the pit under cyclic load was studied using Finite Element Analysis (FEA). The pit geometries used for FEA modelling here are similar in terms of aspect ratio to those generated during experiments. Previous studies have used FEA to study the pit-to-crack transition under static loading using the elastic and elastic-plastic material models [1,10,12–20]. Other than a study by some of the authors of the present work [21], we are not aware of any other published data on finite element analysis of corrosion pits under cyclic loading in air or in environment.

The approach of using empirical studies to generate test data and complementary FE modelling has allowed us to better understand the pit-to-crack damage phenomena. We have reviewed the strain localisation determined by FEA models in the context of the observed crack initiation site(s) on specimens, to propose a fatigue crack initiation process based on the available physical evidence.

2. Methodology

The effect of cyclic loading of 316L specimens containing a single corrosion pit was investigated through corrosion fatigue testing and corresponding FEA. Three pit geometries with different aspect ratios are used in this study. The aspect ratio of the corrosion pits in this work is described in Figure 1 below.

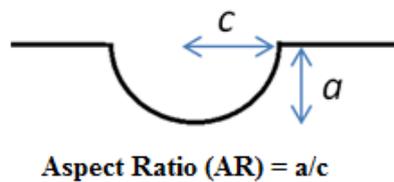


Figure 1: Pit geometry in this work is defined as shown in terms of the Aspect Ratio (AR). AR<1 is termed shallow, AR=1 is symmetrical and AR>1 is considered a deep pit.

2.1. Experiments

2.1.1. Material

The chemical composition of 316L alloy used in this work is given in Table 1. The mechanical properties of the material were derived through tensile testing in accordance with ISO 6892-1 standard. Average values of 230MPa (YS at 0.2% offset) and 595MPa (UTS) were determined.

Table 1: The chemical composition of SS 316L used in this work

Chemical, (% m/m)											
C	Si	Mn	P	S	Cr	Mo	Ni	Al	As	B	Co

0.015	0.40	1.39	0.025	0.003	17.0	2.10	10.1	<0.01	0.008	0.002	0.009
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Cu	Nb	Pb	Sn	Ti	V	W	Ca
0.22	<0.01	0.002	0.010	<0.005	0.02	<0.05	<0.001

Samples were machined using Electrical Discharge Machining (EDM) with dimensions as shown in Figure 2 and in accordance with the requirements of ASTM E466 for fatigue testing.

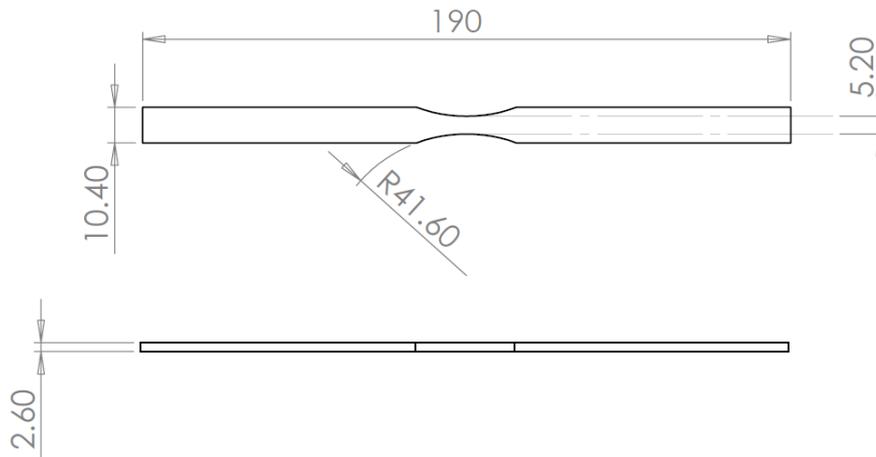


Figure 2: Dimensions, in mm, of the specimens used for fatigue testing

Following the machining process, surfaces of the samples were ground to 1200 grit finish using grit papers and then polished using 6 μ m, 3 μ m and finally 1 μ m synthetic diamond compound. This process ensured a satisfactory surface finish for test specimens, with average surface roughness values (Ra) of 0.095 μ m. Residual stress measurements using X-Ray Diffraction were carried out on the surface of polished samples. Compressive stresses were detected and are presented in Table 2.

Table 2: X-Ray Diffraction surface residual stress measurements of 316L SS surface prepared to Ra 0.095 μ m.

	Surface Residual Stress, MPa	
	Transverse (across the long axis of the specimen)	Longitudinal (along the long axis of the specimen)
Highest	-283.78 \pm 35.08	-402.39 \pm 23.31
Lowest	-237.61 \pm 21.00	-321.45 \pm 21.02

2.1.2. Pre-Pitting

A single corrosion pit was grown on the surface of the specimen at the centre of the gauge length using the micro-capillary electrochemical cell technique [22]. Table 3 shows the dimensions of the three types of pits generated for this study as determined initially by confocal light microscopy and subsequently by X-ray tomography using an Xradia Versa 520 instrument.

Table 3: The dimensions of the single corrosion pits generated for this study

Specimen Number	Diameter, $2c$ (μm)	Depth, a (μm)	Aspect Ratio, a/c
1	780	250	0.64
2	740	340	0.92
3	760	480	1.26

2.1.3. Corrosion Fatigue testing

Following generation of the pitted specimen, corrosion fatigue tests were carried out at room temperature using aerated 3.5wt% NaCl solutions and a Perspex environmental cell, as shown in Figure 3. The fatigue test parameters employed were as follows: stress range ($\Delta\sigma$) of 270 MPa, stress ratio of 0.1 and sinusoidal waveform with frequency 0.5 Hz.

Preliminary corrosion fatigue tests using smooth polished 316L SS specimens (i.e. presenting no corrosion pits or defects) employing same parameters as described above showed no failure up to 10^6 cycles, which defined the fatigue limit to be used in subsequent pit-to-crack tests. Identical test parameters to those used before were selected in order to survey the impact of single corrosion pits on the fatigue performance of pre-pitted 316L SS specimens.

The specimens were run to failure and upon failure, the specimens were removed, rinsed with de-ionised (DI) water and the fracture surfaces investigated using Scanning Electron Microscope (SEM).

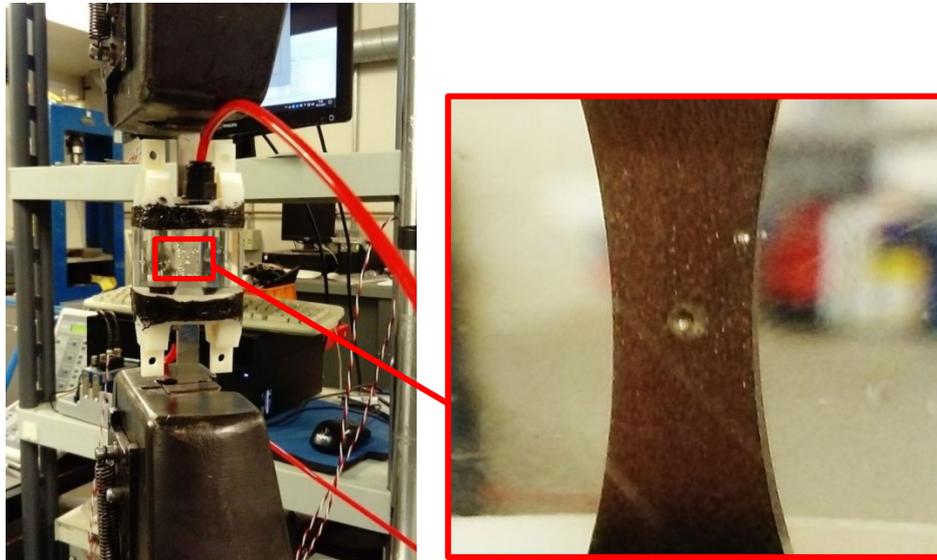


Figure 3 Setup of Corrosion Fatigue testing of specimens containing a single corrosion pit. The cell contains 3.5wt% NaCl.

2.2. Finite Element Analysis (FEA)

Many researchers [1, 9, 11-19] have utilised FEA to study the pit-to-crack transition in different alloys, but to-date all studies have been limited to static loading only. In the present study, in order to consider the effect of cyclic loading and local plastic deformation, 3D elastic-plastic FEA has been carried out using ABAQUS software. The local stress-strain behaviour of idealised (i.e. possessing smooth surfaces) pits of three different dimensions was investigated.

To decrease the computation time, quarter-plate geometry including the pit in the centre of the plate was modelled. The size of the model is shown in Figure 4. Table 4 shows the dimensions of created pits. Ten cycles of a sinusoidal load with the same maximum stress and stress ratio as experiments, i.e. 300MPa and 0.1, were applied using the 3D model. A combined isotropic-kinematic hardening elastic-plastic material model was used by

implementing the cyclic true stress-strain curve of 316L stainless steel, using data provided in the literature [23]. A mesh study was conducted utilising a 10-node second order tetrahedral element (C3D10) to ensure independency of the results with the mesh size. A finer mesh was used around the pit to provide more accurate data in the most important areas. Figure 4 shows the utilised mesh and defined boundary conditions including the symmetry planes.

Table 4: Simulated pit geometries

Pit number	Diameter, $2c$ (μm)	Depth, a (μm)	Aspect Ratio, a/c
1	750	250	0.67
2	750	350	0.93
3	750	450	1.2

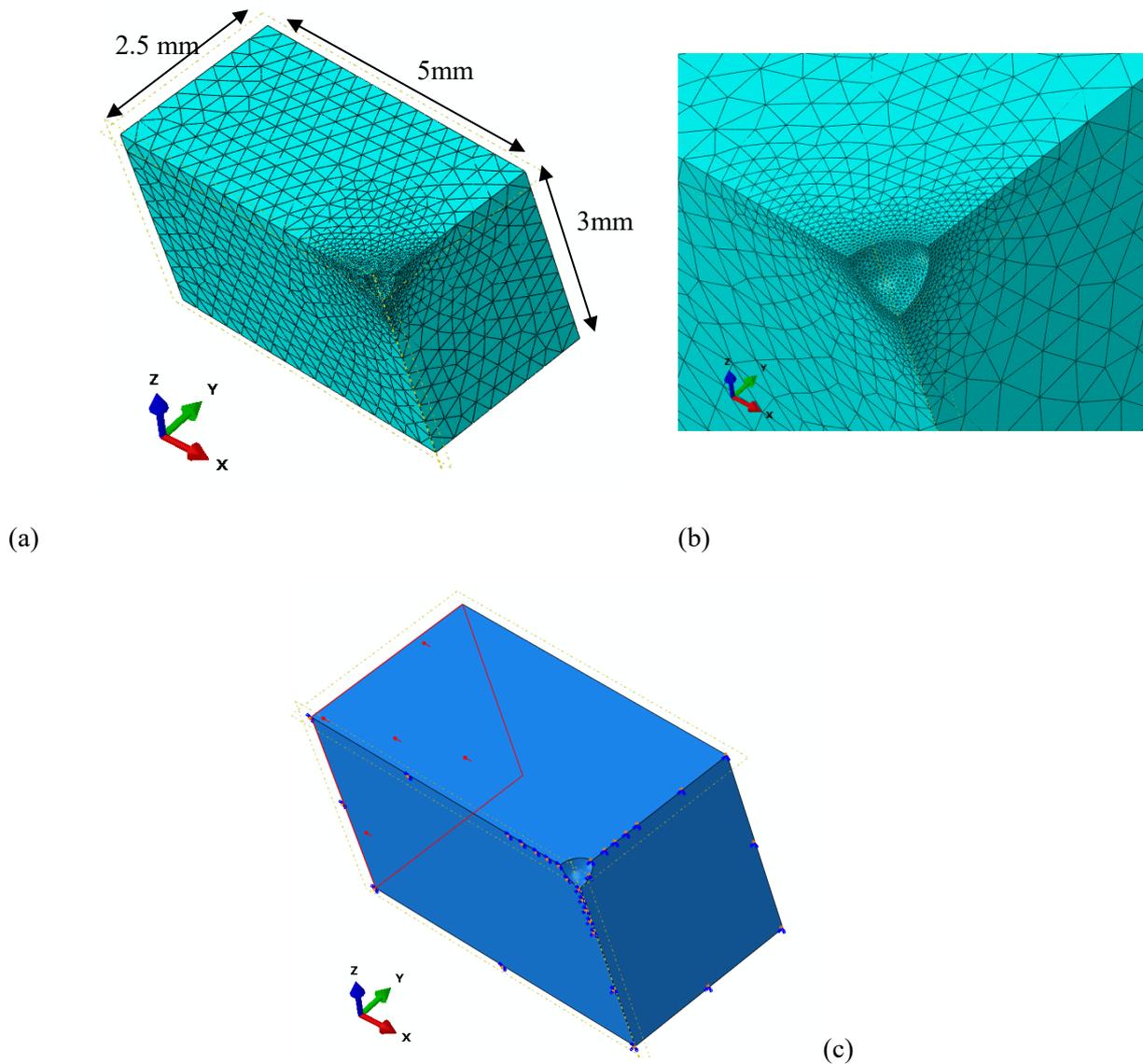


Figure 4: (a) Finite element model and meshing of 1/4 plate containing a pit. (b) More detailed mesh around the local pit area. (c) Defined boundary conditions including planes of symmetry boundary and applied load. X is the loading direction.

3. Results

3.1. Experimental Results

In these corrosion fatigue tests on 316L SS containing single corrosion pits, it is noted that all cracks leading to the final failure of the specimens originated from corrosion pits. Table 5 contains a summary of test results, including the number of cycles to failure for each

specimen containing a single pit with different aspect ratio (Specimen 1; shallow pit; AR = 0.64; Specimen 2; quasi-symmetrical pit; AR = 0.92; Specimen 3; deep pit; AR = 1.26). It is clear that as the aspect ratio of the pit increases, the number of cycles to failure decreases.

Table 5: The number of cycles to failure during corrosion fatigue testing of specimens containing single corrosion pits.

Specimen Number	Diameter, $2c$ (μm)	Depth, a (μm)	Aspect Ratio, a/c	Number of Cycles to failure
1	780	250	0.64	631,357
2	740	340	0.92	496,911
3	760	480	1.26	120,166

Fractography of the failed specimens was carried out using SEM. The analysis showed that cracks initiated:

- at the base of the shallow pit, AR 0.64
- at the shoulder of quasi symmetrical pit, AR 0.92
- at the mouth of the deep pit, AR 1.26

The location of crack initiation in each case is highlighted in Figures 5-7.

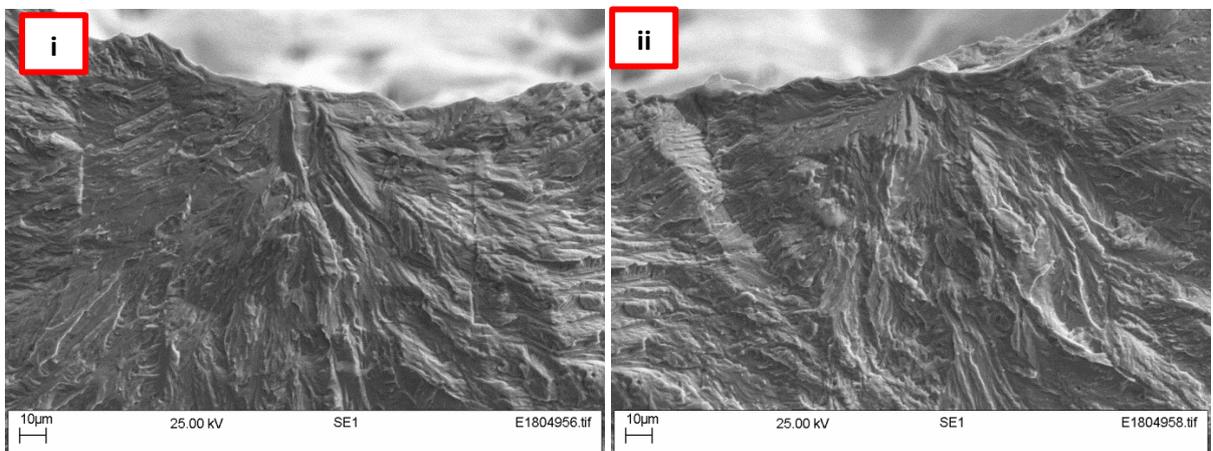
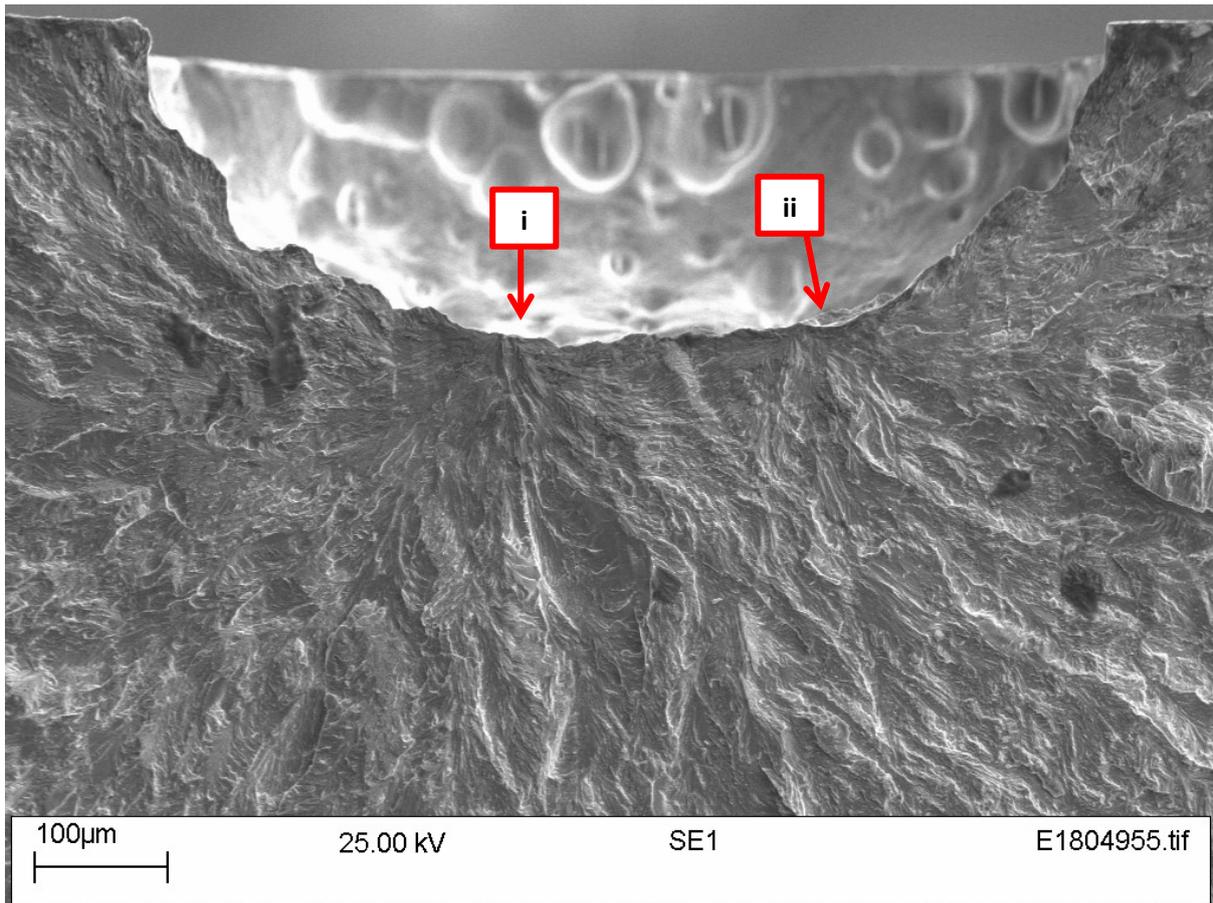


Figure 5: SEM images of crack flank from ‘Specimen 1’ which had a corrosion pit of $AR=0.64$. The images emphasize the two crack initiation sites, i and ii being at the base of the corrosion pit. Higher resolution images of sites i and ii where crack initiated are also provided.

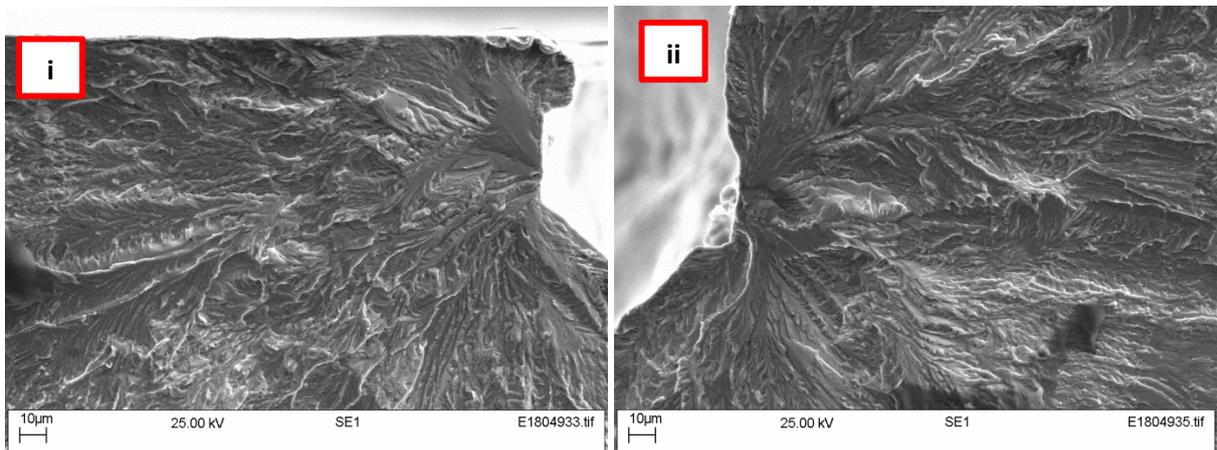
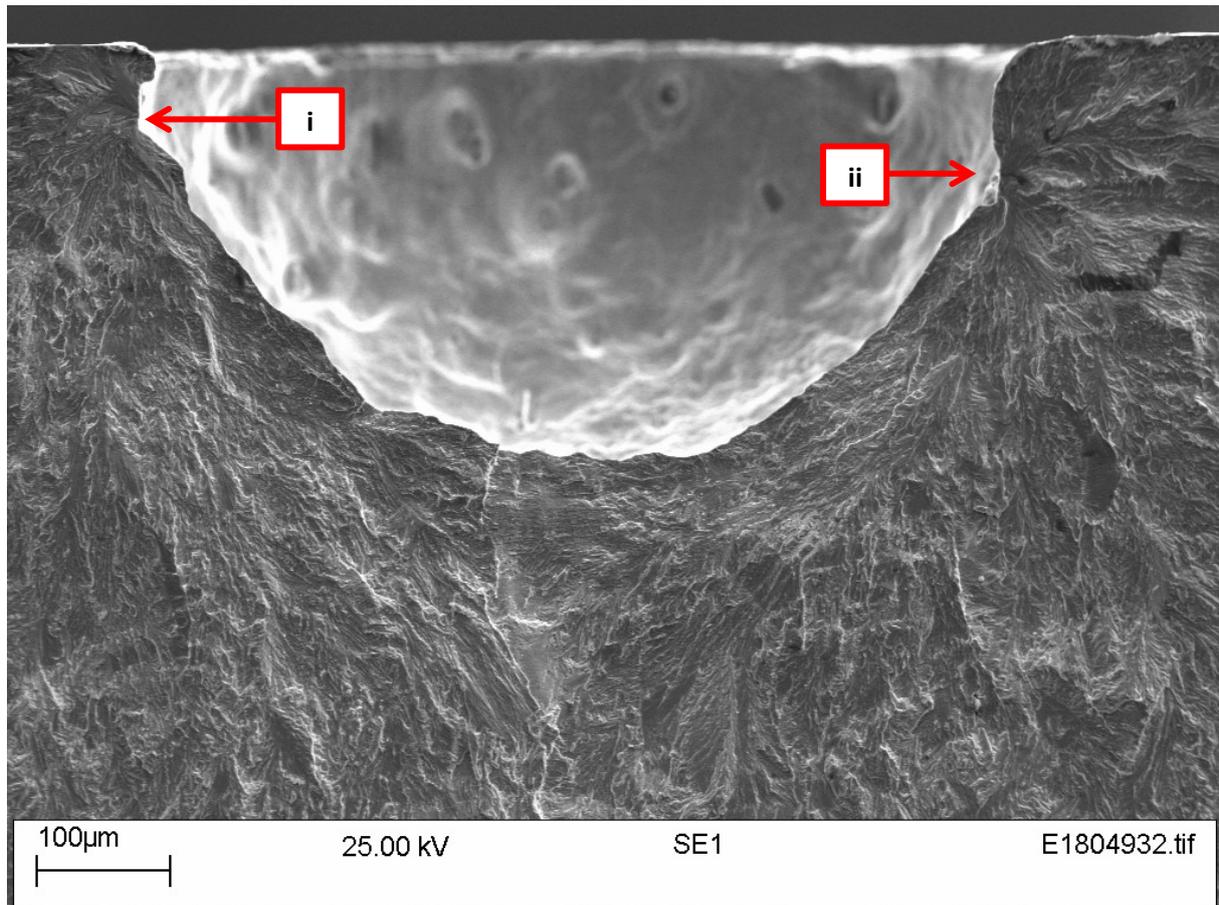


Figure 6: SEM images of crack flank from ‘Specimen 2’ of quasi-symmetrical pit with $AR=0.92$. The images emphasise the two crack initiation sites, i and ii being around the shoulder region of the corrosion pit. Images i and ii are higher resolution images of those sites.

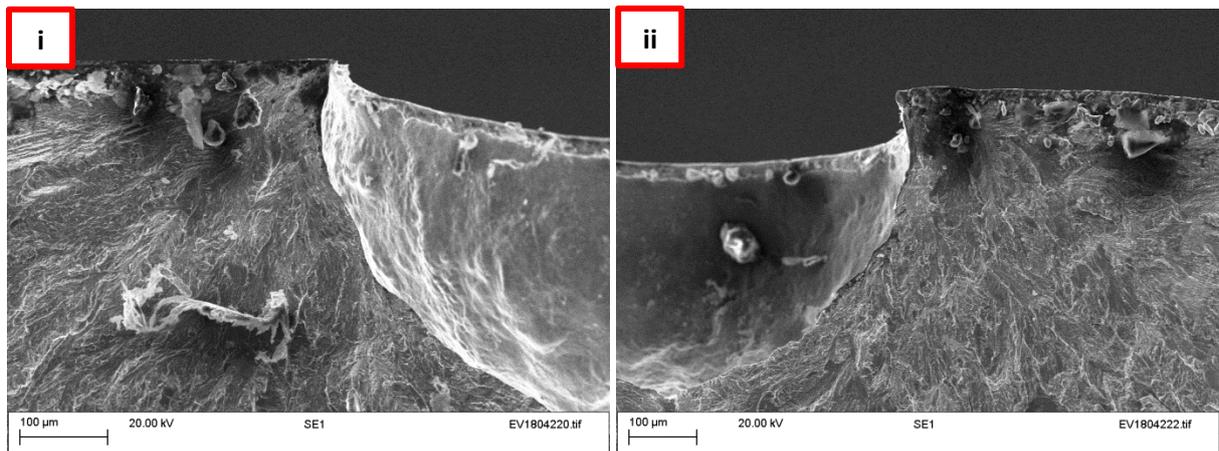
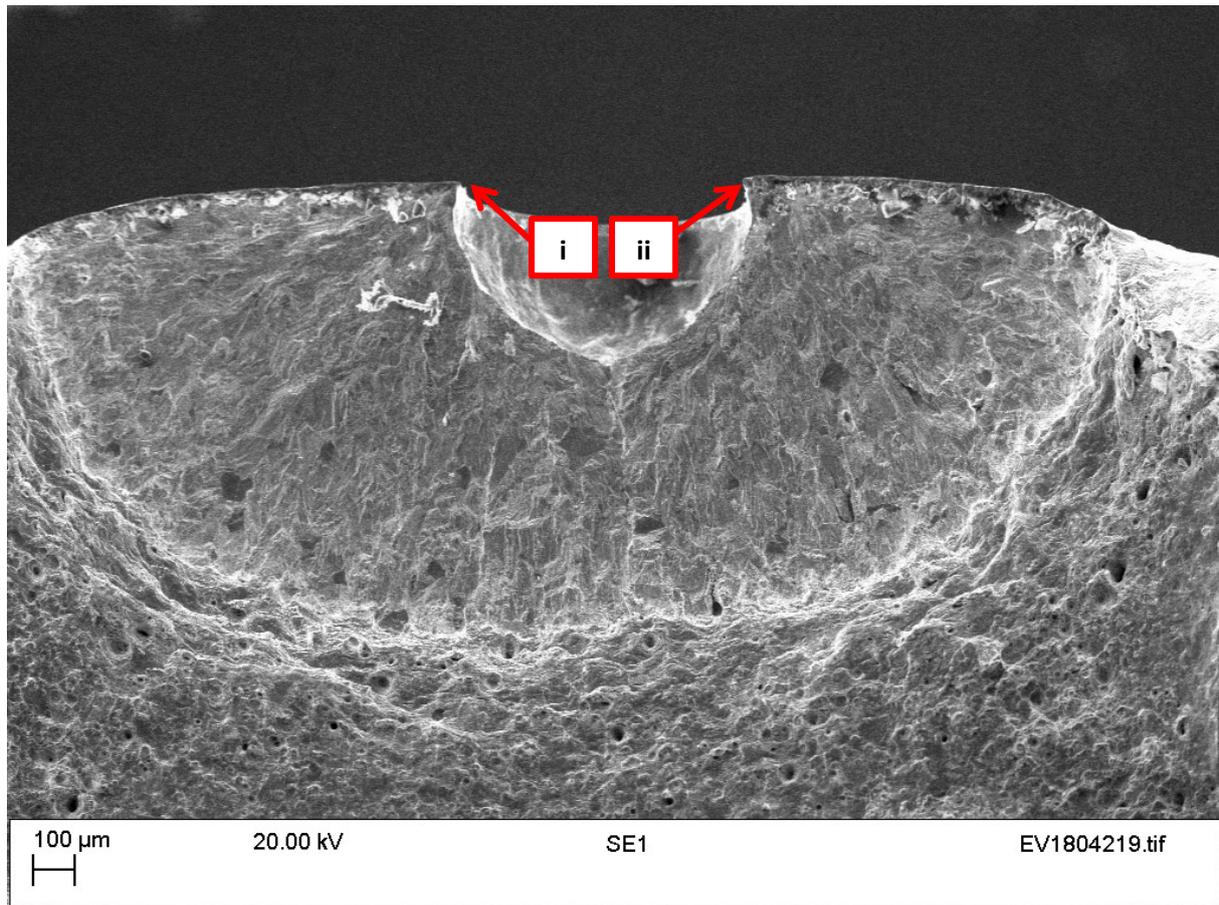
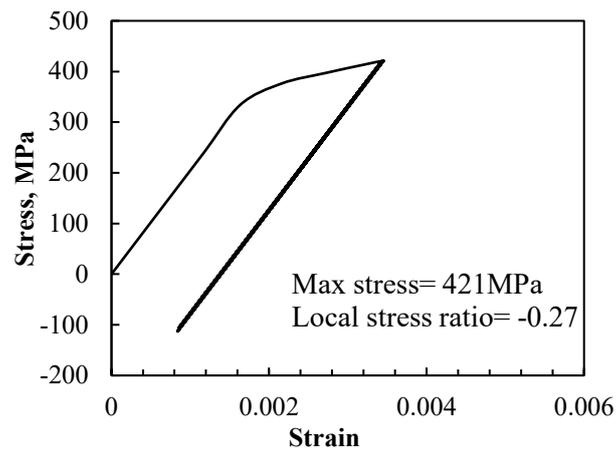


Figure 7: SEM images of crack flank from ‘Specimen 3’ of deep pit with AR=1.26. The images emphasise the two crack initiation sites, i and ii being at the mouth of the corrosion pit. Higher resolution images of sites i and ii where cracks initiated are presented.

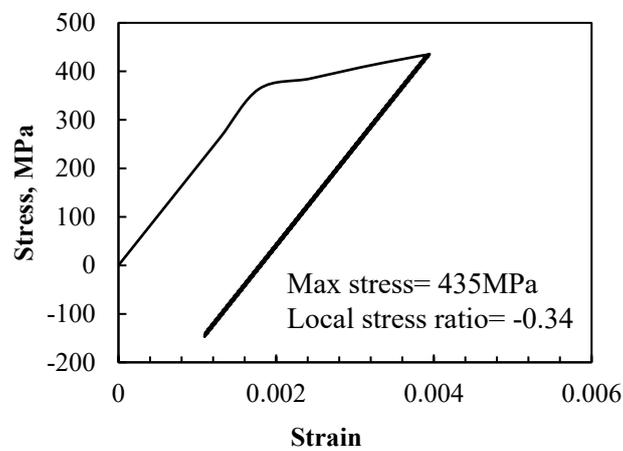
3.2. FEA

3.2.1. *The stress-strain hysteresis loop at the pit base*

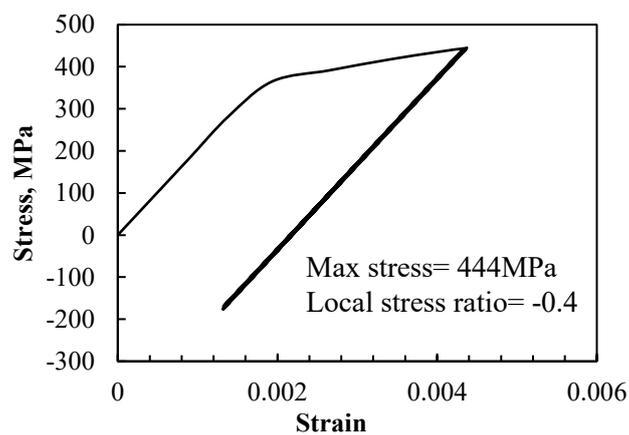
Elastic-plastic FE simulations relating to samples 1-3 (vis. shallow, quasi-symmetrical and deep pit specimens) containing a single central corrosion pit were carried out. Cyclic loading parameters used in the model were identical to those used in tests, i.e. maximum load of 300 MPa and R ratio of 0.1 (i.e. stress range ($\Delta\sigma$) of 270 MPa). The stress-strain hysteresis loop in the loading direction (refer to Figure 4) at the base of the pits was obtained from the results (Figure 8). The maximum stress and local stress ratio is provided on each plot for pits 1-3. As shown in Figure 8, a large plastic strain occurs during the first cycle; thereafter the loops converge upon reaching a stable condition. It is known that multiple rather than single cracks can initiate at the locations of plastic deformation around the pit [24]. Of particular interest is that, although the applied stress ratio is 0.1 (i.e. tensile-tensile load) in all cases, the calculated local stress ratio at the pit base is less than 0.1 and becomes compressive, i.e. decreases from -0.27 to -0.4 with increasing the pit depth. The local stress ratio is shown in each graph. It shows that the entire pit base experienced compressive stress even when the minimum stress of 30MPa was applied to the sample.



(a)



(b)



(c)

Figure 8: Local stress (σ_{xx})-strain (ϵ_{xx}) hysteresis curves, maximum stress and local stress ratio resulted from FEA for the base of pit. Applied stress ratio was 0.1. Pit Aspect Ratio (AR) were: a) 0.67, b) 0.93 and c) 1.2

The pit base maximum stress and strain in each sub figure shows clearly that increasing the pit depth has resulted in a slight increase in the maximum stress at the pit base and a large

change in the maximum strain, i.e. the effect of pit geometry on local maximum stress and strain.

3.2.2 *Stress Distribution*

Figures 9, 10, 11 and 12 show the stress distribution around the three FEA-simulated pits at different time intervals during the first and second loading cycle, i.e. $1 \frac{1}{4}$, $1 \frac{1}{2}$, $1 \frac{3}{4}$ and finally at end of 2nd cycle. For convenience, only quarter sections are reproduced in these images. As evident from the legend in each case, the stress contours indicate negative stress values meaning that despite the applied load being tensile-tensile, the local stress at the pit mouth or the entire cavity is compressive. It is apparent from the figures that more compressive stress is present for higher aspect ratios/pit depth values.

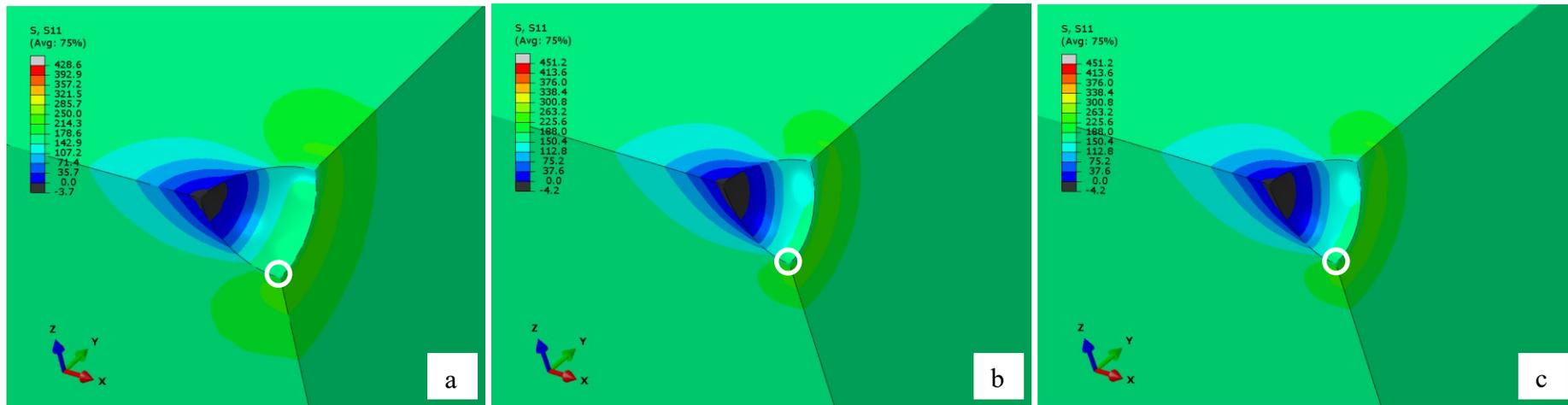


Figure 9 Stress (σ_{xx}) contours after 1 ¼ cycle (loading). The X direction is the loading direction. The pit base is shown by the white circle. (a) Pit AR = 0.67 (b) Pit AR = 0.93 (c) Pit AR = 1.2

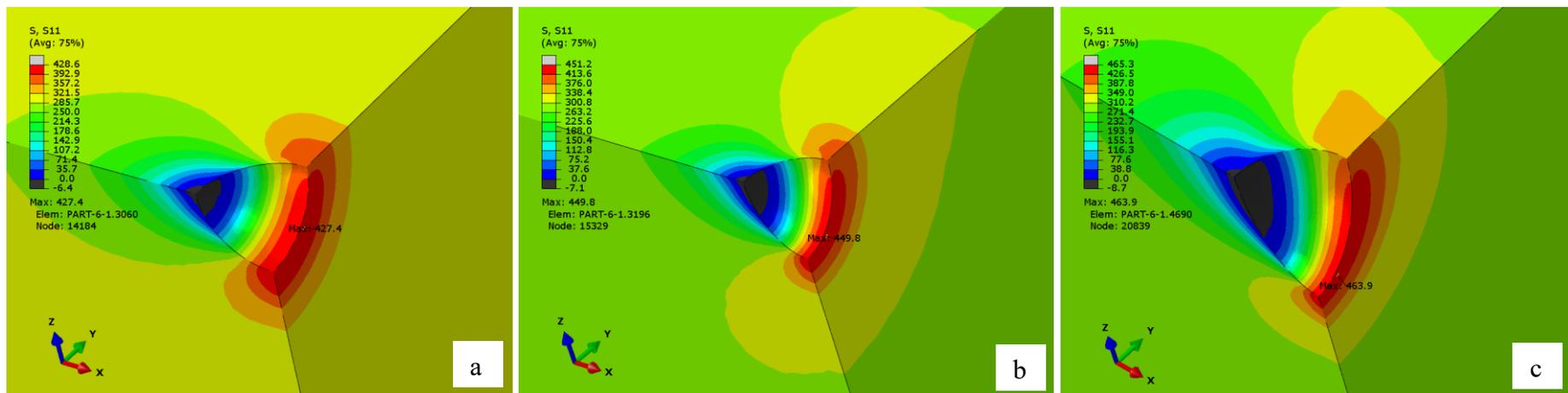


Figure 10 Stress (σ_{xx}) contours after 1 ½ cycle (maximum applied stress). The X direction is the loading direction. (a) Pit AR = 0.67 (b) Pit AR = 0.93 (c) Pit AR = 1.2

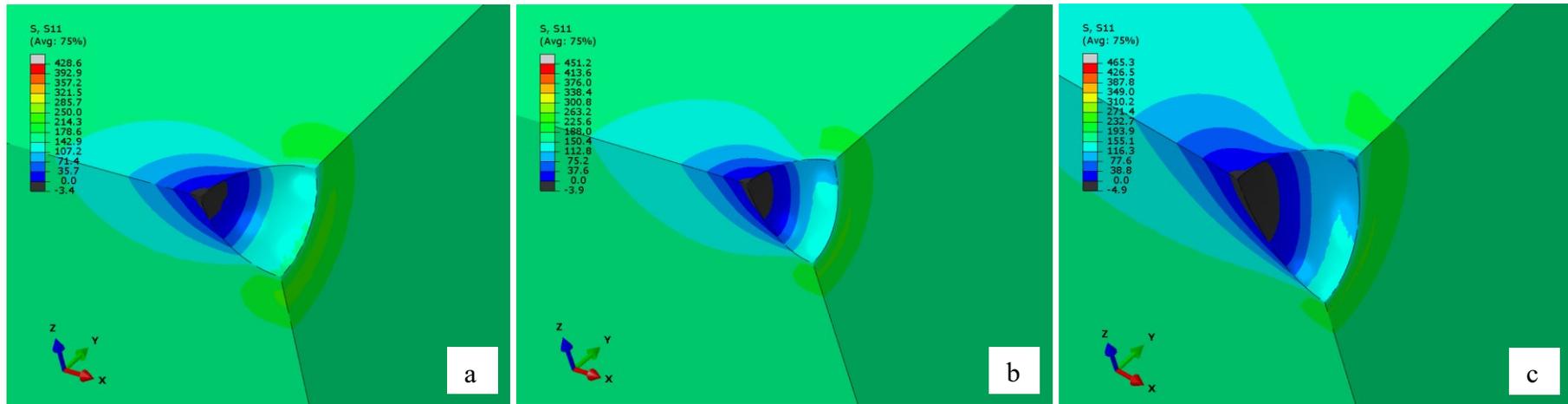


Figure 11 Stress (σ_{xx}) contours after 1 $\frac{3}{4}$ cycle (unloading). The X direction is the loading direction. (a) Pit AR = 0.67 (b) Pit AR = 0.93 (c) Pit AR = 1.2

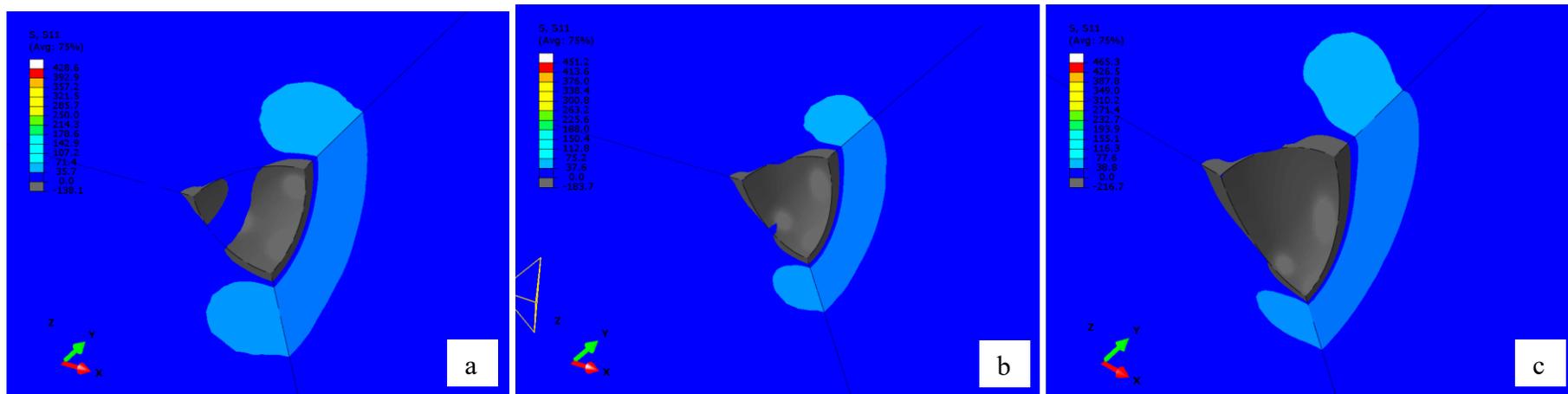


Figure 12 Stress (σ_{xx}) contours after 2nd cycle (minimum applied stress). The X direction is the loading direction. (a) Pit AR = 0.67 (b) Pit AR = 0.93 (c) Pit AR = 1.2

3.2.3. Strain Maps

Strain maps relating to the effect of cyclic loading on the different pit geometries were calculated using the maximum applied load. The results are presented in Figures 13-15. The strain maps suggest fairly uniform strain for low AR of corrosion pits but the strain begins to localise near the shoulder and the mouth region as the AR of the corrosion pit increases (i.e. the corrosion pit is more U shaped). It is interesting to compare the development of strain with that of local stress. The pattern of results suggests that increasing pit aspect ratios tends to relocate the localisation of strain towards the pit mouth. These predictions are in good agreement with the experimental results obtained.

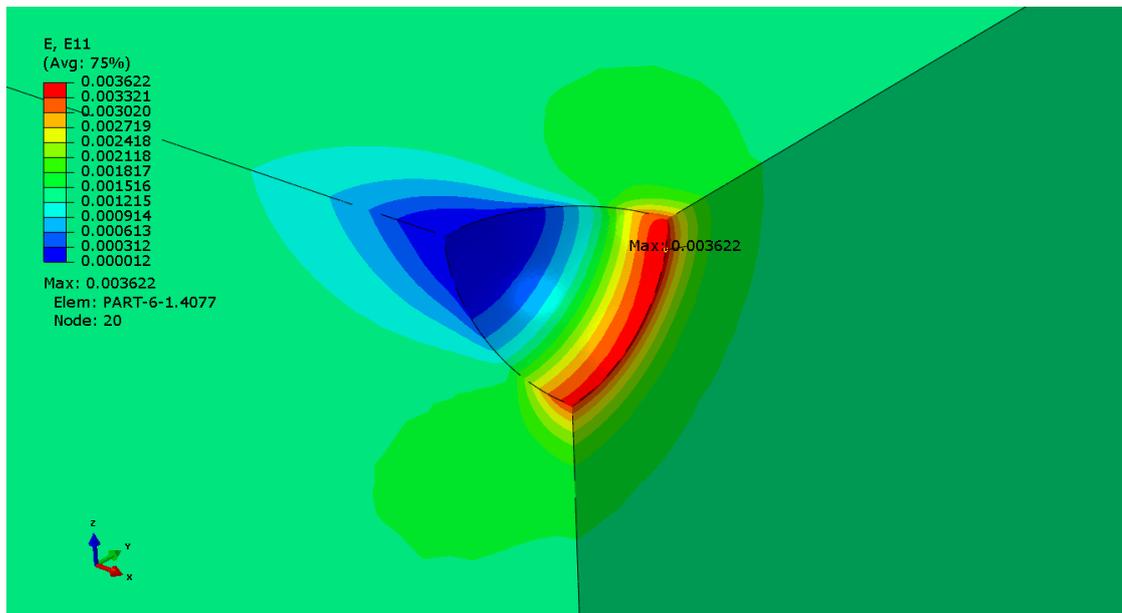


Figure 13: Strain (ϵ_{xx}) map at the maximum applied load for pit with AR 0.67. It shows that the strain is concentrated uniformly around the centre line.

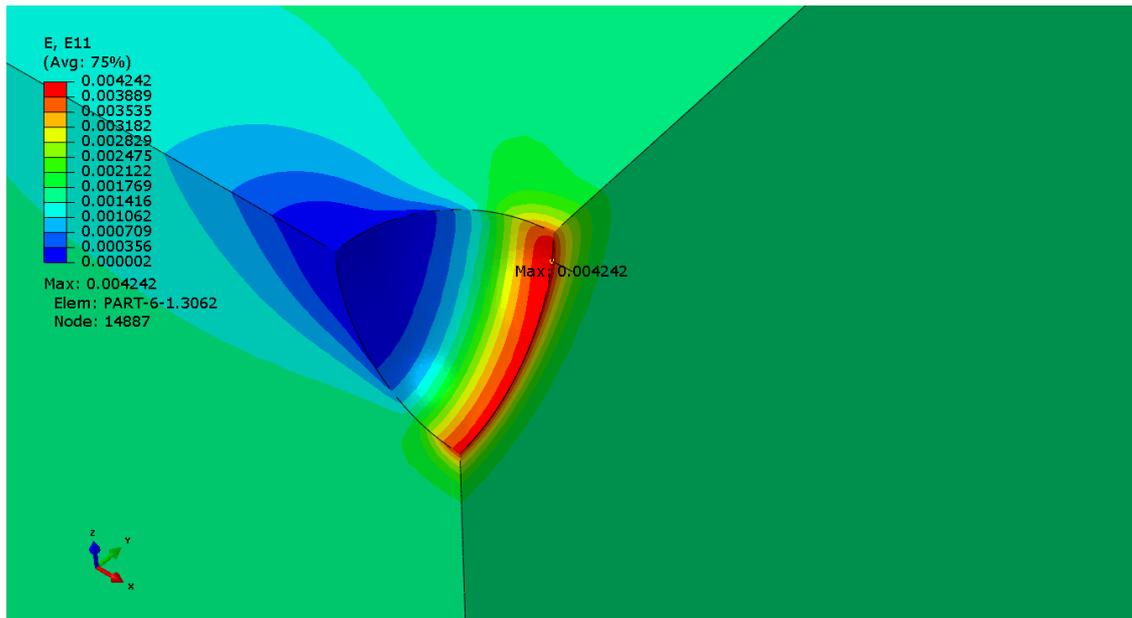


Figure 14: Strain (ϵ_{xx}) map at the maximum applied load for pit with AR 0.92. The strain in this pit is slightly more concentrated from the shoulder region to the mouth of the corrosion pit.

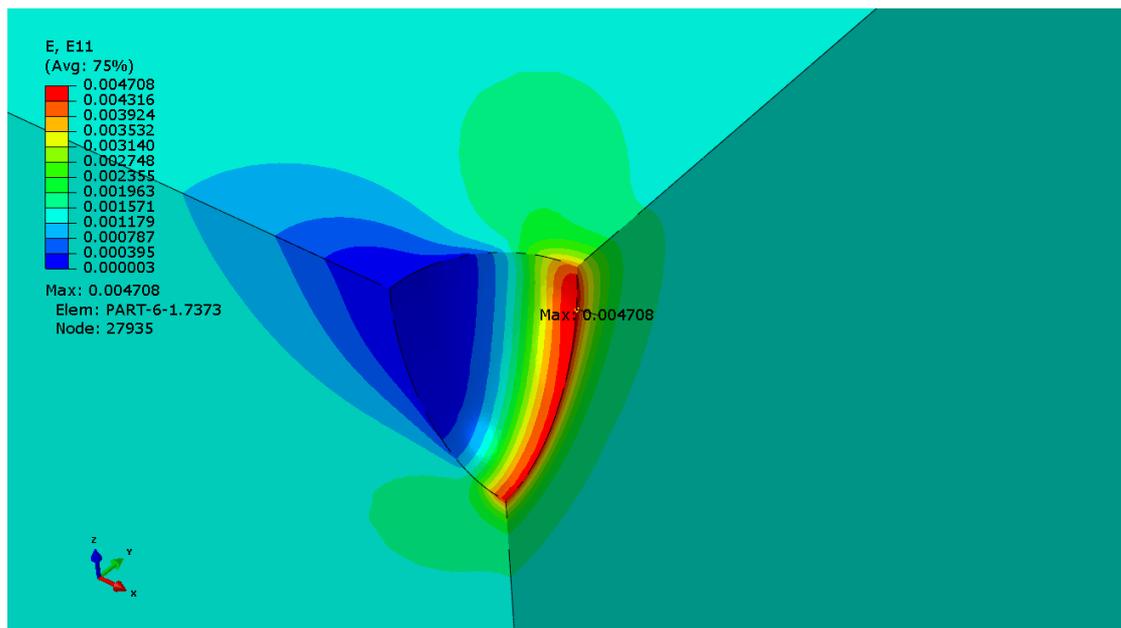


Figure 15: Strain (ϵ_{xx}) map at maximum applied load for pit with AR 1.2. The strain in this pit is significantly more concentrated near the mouth of the pit.

4. Conclusions

The effect of pit geometry and cyclic loading and consequential development of local stress and strains at corrosion pits in stainless steel 316L have been investigated by fatigue testing, fractography examination and finite element analysis. Based on this study, following conclusions can be drawn.

- From the corrosion fatigue tests of a single corrosion pit and fractography analysis, it is seen that the crack initiated at the base of the shallow pit (aspect ratio, a/c , of 0.67, where a = pit depth and c = half-width). The crack initiation site shifted towards pit shoulder or mouth region as the pit depth-to-width aspect ratio is increased. This was further investigated using finite element analysis of the strain localisation around the corrosion pit.
- Finite element analysis has shown that
 - Even when the applied cyclic stress is tension-tension, the local stress in the loading direction at the pit can be tension-compression owing to the plasticity effect. For applied stress ratio of 0.1, the local stress ratio is decreased from -0.27 to -0.4 with increasing pit depth.
 - The magnitude of local compressive stress increases with increasing pit depth. This is due to the increase in the stress concentration factor
 - The strain maps for different pit geometries have shown the following crack initiation trends, which are confirmed by the tests in this study
 - For a shallow pit, with an aspect ratio of 0.64, the strain concentration is uniform around the centre of the pit, suggesting that fatigue crack initiation can be anywhere in that region

- For intermediate pit aspect ratio of 0.93, the strain concentration is near the pit shoulder region; the crack initiation site is more likely to take place in that region.
- For larger pit aspect ratio of 1.2, the strain concentration is near the pit mouth; hence, it is very likely that fatigue crack initiation starts at the pit mouth.

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